



Research article

Photogravitational perturbations in the infinitesimal orbits around the libration points in the oblate RTBP

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Abstract: In this paper, the infinitesimal orbits around the libration points in the photogravitational oblate restricted problem are computed. To reach this goal, the Hamiltonian of our dynamical model taking into account the considered perturbing forces is constructed. A Lie operator method, as a method of solution, is outlined. The Hamiltonian is transferred to any point of the equilibrium point as an origin. The explicit first order as well as the second order solutions for the coordinates and their conjugate momenta of a test particle in an infinitesimal orbit around any equilibrium point are obtained.

Keywords: infinitesimal orbits; RTBP; libration points; a Lie operator; oblateness perturbations; photogravitational perturbations

1. Introduction

The meaning of the infinitesimal orbits are defined as follows: Those orbits that are very close to the equilibrium points. The radii of these orbits are very small. The scientific significance of these orbits come from the fact that the mission designers require to place missions at equilibrium points to have the advantage of these points. The infinitesimal orbits are already used for the mission near any point of the equilibrium points.

The restricted three bodies problem (in brief RTBP) can be investigated directly from the equations of motion given in any textbook of celestial mechanics. Authors can treat this problem

even though with some considered perturbations: e.g., relativistic, photogravitational, dynamical shapes of the primaries, drag,etc. The history of the restricted problem is so long as the beginning of the reviviscence era began with Euler and Lagrange continues with Jacobi, Poincaré and Birkhoff. It continues intensively to the date, one cannot conclusive survey of these works but some relevant works are; Ahmed, et al. [1], Douskos and Perdios [2], Abd El-Salam and Abd El-Bar [3,4], Abd El-Salam and Katour [5], and Abd El-Salam [6]. The Infinitesimal orbits around the equilibrium points in the restricted three body problem (in brief RTBP) are very important for space community. NASA in 1978 launched ISEE3 into a halo orbit at the L_1 of the Earth-Sun system. It was designed to study the Earth-Sun connection through the interaction between the magnetic field of the Earth and the solar wind. In 1996 the SOHO mission was launched to investigate deeply the Sun's internal structure, the solar extensive outer atmosphere and the solar wind origin.

The dynamical behavior of the test particle near the libration points, namely the infinitesimal orbits has been collected in a work by Duncombe and Szebehely [7]. Richardson [8] used successive approximations technique in conjunction with a procedure similar to Poincaré-Lindstedt technique to obtain a 3rd order analytical solution of halo type periodic orbits applied for the Earth-Moon system. Barden and Howell [9] used the central manifold theory to analyze the motions in the vicinity of the collinear equilibrium points. Howell [10] studied the families of orbits in the neighbourhood of the collinear libration points.

Gomez, et al. [11] gave some families of quasi-halo orbits in Sun- (Earth-Moon) and in Earth-Moon systems around L_1 and L_2 . While Gomez, et al. [12] treated the transfer problem between infinitesimal orbits. Selaru and Dimitrescu [13] used asymptotic approximations based on Von Zeipel-type method to study the motions in the vicinity of a equilibrium point in the planar elliptic problem. The eccentricity and inclination effects on small amplitude librations around the triangular points L_4 and L_5 have been studied by Namouni and Murray [14]. The analytical continuation method have been used by Corbera and Llibre [15] to investigate the symmetric periodic orbits around a collinear point in the RTBP. Hamdy, et al. [16] used a perturbation procedure based on Lie series to develop explicit analytical solutions for infinitesimal orbits about the equilibrium points in the elliptic RTBP.

Abd El-Salam [17] studied the periodic orbits around the libration points in the relativistic RTBP. He analysed the elliptic, hyperbolic and degenerate hyperbolic orbits in the vicinity of the L_1 , L_2 and L_3 . He found as well as elliptic orbits in the neighborhood of the L_4 and L_5 .

Ibrahim, et al. [18] presented the special solutions of the RTBP specifying the locations of the equilibrium points. They obtained periodic orbits around these libration points analytically and numerically. Tiwary, et al. [19] described a third-order analytic approximation for computing the three-dimensional periodic halo orbits near the collinear L_1 and L_2 Lagrangian points for the photogravitational circular RTBP in the Sun-Earth system. Tiwary and Kushvah [20] computed halo orbits using Lindstedt-Poincaré method up to fourth order approximation, then analyzed the effects of radiation pressure and oblateness on the orbits around Libration points L_1 and L_2 .

In this work, we will use the Hamiltonian approach to compute the infinitesimal trajectories around the equilibrium points. We will construct first the Hamiltonian of the problem, then it will be followed by outlining of the perturbation procedure used, namely the Delva-Hanselmeir technique, Delva [21] and Hanslmeier [22].

Mittal, et al. [23] have studied periodic orbits generated by Lagrangian solutions of the RTBP when both of the primaries is an oblate body. They have illustrated the periodic orbits for different values of the problem parameters.

Peng, et al. [24] proposed an optimal periodic controller based on continuous low-thrust for the stabilization missions of spacecraft station-keeping and formation-keeping along periodic Libration point orbits of the Sun-Earth system.

Peng, et al. [25] presented the nonlinear closed-loop feedback control strategy for the spacecraft rendezvous problem with finite low thrust between libration orbits in the Sun-Earth system.

Jiang [26] investigated the equilibrium points and orbits around asteroid 1333 Cevenola by considering the full gravitational potential caused by the 3D irregular shape. They calculated gravitational potential and effective potential of asteroid 1333 Cevenola. They also discussed the zero-velocity curves for a massless particle orbiting in the gravitational environment.

Wang [27] applied the developed symplectic moving horizon estimation method to the Earth-Moon L_2 libration point navigation. their numerical simulations demonstrated that though more time-consuming, the proposed method results in better estimation performance than the EKF and the UKF.

We aim to give the explicit formulas for coordinate and momenta of the infinitesimal orbits around one of the libration points in the photogravitational oblate RTBP. The article is organized as follows: In section 1, we gave a brief introduction. While in section 2, we formulated the Hamiltonian in rotating frame of reference. In section 3 we transformed the Hamiltonian near any one of the equilibrium point in the considered model. In section 4 we outline the perturbation approach used. In section 5, and its subsequent subsections we computed the coordinate and momentum vectors of an infinitesimal body revolving one of the equilibrium point in a halo orbit. At the end of the paper we summarize our obtained results.

2. Hamiltonian in inertial frame of reference

The Lagrangian of the problem can be obtained from

$$\mathcal{L} = T - U, \quad (1)$$

where \mathcal{L} is the Lagrangian of the problem, T and U are the kinetic and potential energies of the system respectively, written in terms of the generalised coordinates and velocities and the time ($\mathbf{q} \equiv q_i, \dot{\mathbf{q}} \equiv \dot{q}_i; t$), $i = 1, 2, \dots, n$.

The Legendre transform allows us to switch from the Lagrangian to the Hamiltonian formulism,

$$\mathcal{H} = \sum_{i=1}^n p_i \dot{q}_i - \mathcal{L}(\mathbf{q}, \dot{\mathbf{q}}, t), \quad (2)$$

where the Hamiltonian function $\mathcal{H} \equiv \mathcal{H}(\mathbf{q}, \mathbf{p})$ is a function in the generalized coordinates \mathbf{q} , and the conjugate generalized momenta \mathbf{p} . The Lagrangian describing the motion of the infinitesimal mass in the inertial frame of reference is given by

$$\mathcal{L}_{\text{inertial}} = \frac{1}{2}(\dot{X}^2 + \dot{Y}^2) - U, \quad (3)$$

where \dot{X} and \dot{Y} are the velocity components in the inertial frame of reference of the infinitesimal particle. We will assume the potential energy U of the system is given, in the inertial frame of reference, by

$$U = \frac{q_1(1-\mu)}{((X+\mu)^2+Y^2)^{1/2}} + \frac{q_1A_1(1-\mu)}{2((X+\mu)^2+Y^2)^{3/2}} + \frac{q_2\mu}{((X+\mu-1)^2+Y^2)^{1/2}} + \frac{q_2A_2\mu}{2((X+\mu-1)^2+Y^2)^{3/2}},$$

where A_1 and A_2 denote to the oblateness coefficients of the more and less massive primaries respectively such that $0 < A_i \leq 1$, ($i=1,2$), the respective radiation factors for the massive and less massive primaries are q_i , ($i=1,2$) such that $0 < 1-q_i < 1$, $\mu = m_2/(m_1+m_2)$, $\mu \in (0,1/2)$ is the mass ratio of the less massive body to the total mass of the system, and X, Y are the coordinate components in the inertial frame of reference of the infinitesimal particle.

Since the trajectories of the primaries are given by

$$(X_1, Y_1) = (-\mu \cos nt, -\mu \sin nt), \quad (X_2, Y_2) = ((1-\mu) \cos nt, (1-\mu) \sin nt), \quad (4)$$

where nt , the angle of rotation, is the product mean motion n and time t of the problem, and the location of the infinitesimal body with respect to the primaries in the inertial frame, see Figure 1, is

$$r_1^2 = (X + \mu \cos nt)^2 + (Y + \mu \sin nt)^2, \quad r_2^2 = (X - (1-\mu) \cos nt)^2 + (Y - (1-\mu) \sin nt)^2. \quad (5)$$

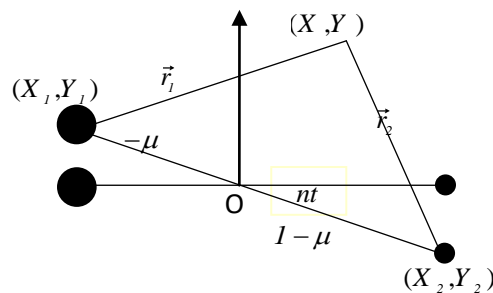


Figure 1. The location of the infinitesimal body with respect to the primaries in the inertial frame.

The amended potential U and hence the inertial Lagrangian $\mathcal{L}_{\text{inertial}}$ is a time independent. To have a clearer insight into many behaviours of the RTBP especially the motion near the Lagrangian points, transform to a rotating system with coordinates ξ and η using

$$\left. \begin{aligned} X &= \xi \cos nt - \eta \sin nt, \\ Y &= \xi \sin nt + \eta \cos nt, \end{aligned} \right\} \quad (6)$$

and the the corresponding velocities transform as

$$\left. \begin{aligned} \dot{X} &= \dot{\xi} \cos nt - \dot{\eta} \sin nt - \xi n \sin nt - \eta n \cos nt, \\ \dot{Y} &= \dot{\xi} \sin nt + \dot{\eta} \cos nt + \xi n \cos nt - \eta n \sin nt, \end{aligned} \right\} \quad (7)$$

where $\dot{\xi}$ and $\dot{\eta}$ are the velocity components in the rotating system. The distances become

$$r_1^2 = (\xi + \mu)^2 + \eta^2, \quad r_2^2 = (\xi - (1 - \mu))^2 + \eta^2, \quad (8)$$

and the new Lagrangian is

$$\mathcal{L}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta}) = \frac{1}{2}(\dot{\xi} - n\eta)^2 + \frac{1}{2}(\dot{\eta} + n\xi)^2 - U, \quad (9)$$

Where the amended potential U is given by

$$\begin{aligned} U &= \frac{n}{2}(\xi^2 + \eta^2) + \frac{q_1(1-\mu)}{((\xi + \mu)^2 + \eta^2)^{1/2}} + \frac{q_1 A_1(1-\mu)}{2((\xi + \mu)^2 + \eta^2)^{3/2}} \\ &+ \frac{q_2 \mu}{((\xi + \mu - 1)^2 + \eta^2)^{1/2}} + \frac{q_2 A_2 \mu}{2((\xi + \mu - 1)^2 + \eta^2)^{3/2}}. \end{aligned} \quad (10)$$

The distances of the infinitesimal mass from the barycenter is given by $r = \sqrt{\xi^2 + \eta^2}$, and n is given by

$$n^2 = 1 + \frac{3}{2}A_1 + \frac{3}{2}A_2.$$

Now, to formulate the Hamiltonian in terms of the generalized coordinates $\mathbf{q} \equiv (\xi, \eta)$ and their canonical conjugate momenta $\mathbf{p} \equiv (p_\xi, p_\eta)$, the following relations are required

$$p_\xi = \frac{\partial \mathcal{L}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta})}{\partial \dot{\xi}}, \quad p_\eta = \frac{\partial \mathcal{L}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta})}{\partial \dot{\eta}}. \quad (11)$$

From Eqs. (9) and (10), we can obtain

$$p_\xi = \dot{\xi} - n\eta, \quad (12)$$

$$p_\eta = \dot{\eta} + n\xi, \quad (13)$$

solution for $\dot{\xi}$ and $\dot{\eta}$ yields

$$\dot{\xi} = p_\xi + n\eta, \quad (14)$$

$$\dot{\eta} = p_{\eta} - n\xi. \quad (15)$$

The Hamiltonian in the rotating frame is given as

$$\mathcal{H}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta}) = \dot{\xi} p_{\xi} + \dot{\eta} p_{\eta} - \mathcal{L}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta}), \quad (16)$$

or, using Eq. (15),

$$\mathcal{H}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta}) = p_{\xi}(p_{\xi} + n\eta) + p_{\eta}(p_{\eta} - n\xi) - \mathcal{L}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta}). \quad (17)$$

Using Eqs. (9), (10) and (15), the Hamiltonian of the problem in terms of the generalized coordinates $\mathbf{q} \equiv (\xi, \eta)$ and the generalized momenta $\mathbf{p} \equiv (p_{\xi}, p_{\eta})$ can be written in the form

$$\begin{aligned} \mathcal{H}_{\text{rotating}}(\xi, \eta, \dot{\xi}, \dot{\eta}) = & \frac{1}{2} p_{\xi}^2 + \frac{1}{2} p_{\eta}^2 + n\eta p_{\xi} - n\xi p_{\eta} + \frac{n}{2}(\xi^2 + \eta^2) + \frac{q_1(1-\mu)}{((\xi + \mu)^2 + \eta^2)^{1/2}} \\ & + \frac{q_1 A_1 (1-\mu)}{2((\xi + \mu)^2 + \eta^2)^{3/2}} + \frac{q_2 \mu}{((\xi + \mu - 1)^2 + \eta^2)^{1/2}} + \frac{q_2 A_2 \mu}{2((\xi + \mu - 1)^2 + \eta^2)^{3/2}}. \end{aligned} \quad (18)$$

3. Transformed Hamiltonian near any equilibrium point

At this point we are interested in the infinitesimal orbits near any equilibrium point. Moving the origin to any point of equilibrium and denoting to the new coordinates and momenta be $(x_1, x_2, P_{x_1}, P_{x_2})$, then from the geometry illustrated in Figure 2, we have

$$\vec{r} = \vec{r}' + \vec{r}_{L_i}, \quad (19)$$

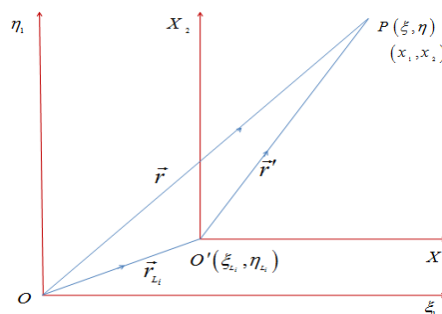


Figure 2. The location of the infinitesimal body with respect to the primaries in the inertial frame.

from Eq. (19)

$$\xi = x_1 + \xi_{L_i}, \quad \eta = x_2 + \eta_{L_i}, \quad i = 1, 2, 3, 4, 5 \quad (20)$$

where (ξ_{l_i}, η_{l_i}) , $i = 1, 2, 3, 4, 5$ are the locations of the equilibrium points, given by Abd El-Salam, et al. [28], Abd El-Salam and Abd El-Bar [29,30], disregarding the relativistic effects in these works. The new momenta read

$$p_\xi = P_{x_1} - n\eta_{l_i}, \quad p_\eta = P_{x_2} + n\xi_{l_i}. \quad (21)$$

Now to obtain the Hamiltonian in the new coordinates, we substitute Eqs. (20) and (21) into Eq. (18) :

$$\begin{aligned} \mathcal{H}(x_1, x_2, P_{x_1}, P_{x_2}) = & \frac{1}{2}(P_{x_1} + nx_2)^2 + \frac{1}{2}(P_{x_2} - nx_1)^2 + q_1(1-\mu)S_1 \\ & + \frac{1}{2}q_1A_1(1-\mu)S_1^3 + q_2\mu S_2 + \frac{1}{2}q_2A_2\mu S_2^3, \end{aligned} \quad (22)$$

where

$$\left. \begin{aligned} S_1 &= \frac{1}{\sqrt{(x_1 + \xi_{l_i} + \mu)^2 + (x_2 + \eta_{l_i})^2}}, \\ S_2 &= \frac{1}{\sqrt{(x_1 + \xi_{l_i} + \mu - 1)^2 + (x_2 + \eta_{l_i})^2}}. \end{aligned} \right\} \quad (23)$$

4. Perturbation approach and solutions

We utilize an approach developed by Delva [21], and Hanslmeier [22]. They carried out the procedure with a differential operator D , the Lie operator, which is a special linear operator that produces a Lie series. The convergence of this latter series is the same as Taylor series, It merely represents another form of the Taylor series whose terms are generated by the Lie operator. We will use the Lie series form for two reasons. The first reason is: The requirement to build up a perturbative scheme at different orders of the orbital elements. The second reason is: Its usefulness also in treating the non-autonomous system of differential equations and non-canonical systems. This enables a rapid successive calculation of the orbit. In addition we can arbitrarily choose the stepsize easily (if necessary). This is an important advantage for the treatment of the problems which has a variable stepsize, e.g., for the mass change of the primaries. The formulas has an easy analytical structure and may be programmed without difficulty and without imposing extra conditions on the convergence. The iteration can be used to generate any desired order of solution, the series can be continued up to any satisfactory convergence reached. The Lie operator is defined as

$$D\Xi = \sum_{i=1}^2 \left(\frac{\partial \Xi}{\partial x_i} \frac{dx_i}{dt} + \frac{\partial \Xi}{\partial P_{x_i}} \frac{dP_{x_i}}{dt} \right) + \frac{\partial}{\partial t}, \quad \Xi = \Xi(x(x_i, P_{x_i}), \mathbf{P}(x_i, P_{x_i})), \quad (24)$$

Leibnitz formula can be used for computing the n^{th} derivative of a product, as

$$\frac{d^n}{dZ^n} [g(z)h(z)] = \sum_{m=0}^n C_m^n \frac{d^m g}{dZ^m} \frac{d^{n-m} h}{dZ^{n-m}}, \quad C_m^n = \frac{n!}{m!(n-m)!}. \quad (25)$$

The n^{th} application of the Lie operator denoted by $D^{(n)}$ takes the form

$$D^{(n)}\Xi = \sum_{i=1}^2 \sum_{m=0}^n C_m^n \left[\left(\frac{\partial^m \Xi}{\partial x_i^m} \frac{d^{n-m} x_i}{dt^{n-m}} + \frac{\partial^m \Xi}{\partial P_{x_i}^m} \frac{d^{n-m} P_{x_i}}{dt^{n-m}} \right) + \frac{\partial^n \Xi}{\partial t^n} \right]. \quad (26)$$

Now using the canonical equations of motion

$$\frac{dx_i}{dt} = \frac{\partial \mathcal{H}}{\partial P_{x_i}}, \quad \frac{dP_{x_i}}{dt} = -\frac{\partial \mathcal{H}}{\partial x_i},$$

we can evaluate the derivatives $\frac{d^{n-m} x_i}{dt^{n-m}}$, $\frac{d^{n-m} P_{x_i}}{dt^{n-m}}$ then we can reach to the solutions (coordinate and momentum vectors, \mathbf{x} and \mathbf{P} respectively) as;

$$\left. \begin{aligned} \mathbf{x} &= (e^{(t-t_0)D})\mathbf{x} \Big|_{x=x_0} = \sum_{j=0}^{\infty} \frac{(t-t_0)^j}{j!} D^{(j)}\mathbf{x} \Big|_{x=x_0, \mathbf{P}=\mathbf{P}_0} = \sum_{j=0}^{\infty} \frac{(t-t_0)^j}{j!} \{D^{(j)}x_1 + D^{(j)}x_2\} \Big|_{x=x_0, \mathbf{P}=\mathbf{P}_0}, \\ \mathbf{P} &= (e^{(t-t_0)D})\mathbf{P} \Big|_{x=x_0, \mathbf{P}=\mathbf{P}_0} = \sum_{j=0}^{\infty} \frac{(t-t_0)^j}{j!} D^{(j)}\mathbf{P} \Big|_{x=x_0, \mathbf{P}=\mathbf{P}_0} = \sum_{j=0}^{\infty} \frac{(t-t_0)^j}{j!} \{D^{(j)}P_{x_1} + D^{(j)}P_{x_2}\} \Big|_{x=x_0, \mathbf{P}=\mathbf{P}_0}. \end{aligned} \right\} \quad (27)$$

As is clear from Eq. (27), the applications of the Lie operator $D^j \Xi$ at different orders are evaluated for the initial conditions of the canonical elements.

5. Solutions at different orders

In this section we are going to evaluate the solutions at different orders. From the definition of the operator $D^{(n)}$, Eq. (26), we get the following explicit expressions at different orders as follows.

5.1. The first order solution

Setting $n = 1$ in Eq. (26) we obtain the required coefficients in Eq. (27) to yield the first order. The required partial derivatives can be obtained using Eq. (22) as follows

$$\begin{aligned} \frac{\partial \mathcal{H}}{\partial x_1} &= -n(P_{x_2} - nx_1) - q_1(1-\mu)(x_1 + \xi_{L_i} + \mu)S_1^3 - \frac{3}{2}q_1A_1(1-\mu)(x_1 + \xi_{L_i} + \mu)S_1^5 \\ &\quad - q_2\mu(x_1 + \xi_{L_i} + \mu - 1)S_2^3 - \frac{3}{2}q_2A_2\mu(x_1 + \xi_{L_i} + \mu - 1)S_2^5, \end{aligned} \quad (28.1)$$

$$\frac{\partial \mathcal{H}}{\partial x_2} = n(P_{x_1} + nx_2) - q_1(1-\mu)(x_2 + \eta_{L_i})S_1^3 - \frac{3}{2}q_1A_1(1-\mu)(x_2 + \eta_{L_i})S_1^5$$

$$-q_2\mu(x_2 + \eta_{L_1})S_2^3 - \frac{3}{2}q_2A_2\mu(x_2 + \eta_{L_1})S_2^5, \quad (28.2)$$

$$\frac{\partial \mathcal{H}}{\partial P_{x_1}} = (P_{x_1} + nx_2), \quad (28.3)$$

$$\frac{\partial \mathcal{H}}{\partial P_{x_2}} = (P_{x_2} - nx_1). \quad (28.4)$$

Substituting Eqs. (28.1)–(28.4) into Eq. (26) and neglecting the very small magnitude terms yields

$$\left. \begin{aligned} D^{(0)}x_1 &= \sum_{i=0}^2 \mathcal{J}_i^1 \mu^i, \\ D^{(0)}x_2 &= \sum_{i=0}^2 \mathcal{N}_i^1 \mu^i, \\ D^{(0)}P_{x_1} &= \sum_{i=0}^2 \mathcal{K}_i^1 \mu^i, \\ D^{(0)}P_{x_2} &= \sum_{i=0}^2 \mathcal{G}_i^1 \mu^i. \end{aligned} \right\} \quad (29)$$

Where the nonvanishing included coefficients are given by

$$\begin{aligned} \mathcal{J}_0^1 &= x_1 + P_{x_1}x_1 - nP_{x_1}x_1 + P_{x_2}x_1 + nP_{x_2}x_1 - nx_1^2 \\ &\quad + q_1S_1^3x_1^2 + 3A_1q_1S_1^5x_1^2 - n^2x_1^2 + nx_1x_2 + q_1S_1^3x_1x_2 + (3/2)A_1q_1S_1^5x_1x_2 \\ &\quad - n^2x_1x_2 + (3/2)A_1q_1S_1^5x_1\eta_{L_1} + q_1S_1^3x_1\xi_{L_1} + q_1S_1^3x_1\eta_{L_1} + 3A_1q_1S_1^5x_1\xi_{L_1} \\ \mathcal{J}_1^1 &= q_1S_1^3x_1 + 3A_1q_1S_1^5x_1 - q_2S_2^3x_1 - 3A_2q_2S_2^5x_1 - q_1S_1^3x_1^2 - 3A_1q_1S_1^5x_1^2 \\ &\quad + q_2S_2^3x_1^2 + 3A_2q_2S_2^5x_1^2 - q_1S_1^3x_1x_2 - (3/2)A_1q_1S_1^5x_1x_2 + q_2S_2^3x_1x_2 \\ &\quad + (3/2)A_2q_2S_2^5x_1x_2 - q_1S_1^3x_1\eta_{L_1} - (3/2)A_1q_1S_1^5x_1\eta_{L_1} + (3/2)A_2q_2S_2^5x_1\eta_{L_1} \\ &\quad + 3A_2q_2S_2^5x_1\xi_{L_1} - q_1S_1^3x_1\xi_{L_1} - 3A_1q_1S_1^5x_1\xi_{L_1} + q_2S_2^3x_1\xi_{L_1} + q_2S_2^3x_1\eta_{L_1} \\ \mathcal{J}_2^1 &= -q_1S_1^3x_1 - 3A_1q_1S_1^5x_1 + q_2S_2^3x_1 + 3A_2q_2S_2^5x_1 \\ \mathcal{N}_0^1 &= x_2 + P_{x_1}x_2 - nP_{x_1}x_2 + P_{x_2}x_2 + nP_{x_2}x_2 - nx_1x_2 - n^2x_1x_2 \\ &\quad + q_1S_1^3x_1x_2 + 3A_1q_1S_1^5x_1x_2 + nx_2^2 - n^2x_2^2 + q_1S_1^3x_2^2 + (3/2)A_1q_1S_1^5x_2^2 \\ &\quad + q_1S_1^3x_2\eta_{L_1} + (3/2)A_1q_1S_1^5x_2\eta_{L_1} + q_1S_1^3x_2\xi_{L_1} + 3A_1q_1S_1^5x_2\xi_{L_1} \\ \mathcal{N}_1^1 &= q_1S_1^3x_2 - q_2S_2^3x_2 - q_1S_1^3x_1x_2 + q_2S_2^3x_2^2 + q_2S_2^3x_1x_2 + 3A_2q_2S_2^5x_1x_2 - q_1S_1^3x_2^2 \end{aligned}$$

$$\begin{aligned}
& -(3/2)A_1q_1S_1^5x_2^2 + (3/2)A_2q_2S_2^5x_2^2 - q_1S_1^3x_2\eta_{l_4} - (3/2)A_1q_1S_1^5x_2\eta_{l_4} + q_2S_2^3x_2\eta_{l_4} \\
& - 3A_2q_2S_2^5x_2 + (3/2)A_2q_2S_2^5x_2\eta_{l_4} - q_1S_1^3x_2\xi_{l_4} - 3A_1q_1S_1^5x_2\xi_{l_4} + q_2S_2^3x_2\xi_{l_4} + 3A_2q_2S_2^5x_2\xi_{l_4} \\
\mathcal{N}_2^1 = & -3A_1q_1S_1^5x_2 - q_1S_1^3x_2 + q_2S_2^3x_2 + 3A_2q_2S_2^5x_2 \\
\mathcal{K}_0^1 = & P_{x_1} + P_{x_1}^2 - nP_{x_1}^2 + P_{x_1}P_{x_2} + nP_{x_1}P_{x_2} - nP_{x_1}x_1 - n^2P_{x_1}x_1 + P_{x_1}q_1S_1^3x_1 \\
& + 3A_1P_{x_1}q_1S_1^5x_1 + nP_{x_1}x_2 - n^2P_{x_1}x_2 + P_{x_1}q_1S_1^3x_2 + (3/2)A_1P_{x_1}q_1S_1^5x_2 \\
& + (3/2)A_1P_{x_1}q_1S_1^5\eta_{l_4} + P_{x_1}q_1S_1^3\xi_{l_4} + 3A_1P_{x_1}q_1S_1^5\xi_{l_4} + 3P_{x_1}\xi_{l_4} \\
\mathcal{K}_1^1 = & P_{x_1}q_1S_1^3 + 3A_1P_{x_1}q_1S_1^5 - P_{x_1}q_2S_2^3 - 3A_2P_{x_1}q_2S_2^5 - P_{x_1}q_1S_1^3x_1 - 3A_1P_{x_1}q_1S_1^5x_1 \\
& + P_{x_1}q_2S_2^3x_1 + 3A_2P_{x_1}q_2S_2^5x_1 - P_{x_1}q_1S_1^3x_2 - (3/2)A_1P_{x_1}q_1S_1^5x_2 + P_{x_1}q_2S_2^3x_2 \\
& + (3/2)A_2P_{x_1}q_2S_2^5x_2 - P_{x_1}q_1S_1^3\eta_{l_4} - (3/2)A_1P_{x_1}q_1S_1^5\eta_{l_4} + P_{x_1}q_2S_2^3\eta_{l_4} \\
& + (3/2)A_2P_{x_1}q_2S_2^5\eta_{l_4} - P_{x_1}q_1S_1^3\xi_{l_4} - 3A_1P_{x_1}q_1S_1^5\xi_{l_4} + 3A_2P_{x_1}q_2S_2^5\xi_{l_4} \\
\mathcal{K}_2^1 = & -P_{x_1}q_1S_1^3 - 3A_1P_{x_1}q_1S_1^5 + P_{x_1}q_2S_2^3 + 3A_2P_{x_1}q_2S_2^5, \\
\mathcal{G}_0^1 = & P_{x_2} + P_{x_1}P_{x_2} - nP_{x_1}P_{x_2} + P_{x_2}^2 + nP_{x_2}^2 - nP_{x_2}x_1 - n^2P_{x_2}x_1 + P_{x_2}q_1S_1^3x_1 \\
& + 3A_1P_{x_2}q_1S_1^5x_1 + nP_{x_2}x_2 - n^2P_{x_2}x_2 + P_{x_2}q_1S_1^3x_2 + (3/2)A_1P_{x_2}q_1S_1^5x_2 \\
& - P_{x_2}\eta_{l_4}^3 + P_{x_2}q_1S_1^3\eta_{l_4} + (3/2)A_1P_{x_2}q_1S_1^5\eta_{l_4} + P_{x_2}q_1S_1^3\xi_{l_4} + 3A_1P_{x_2}q_1S_1^5\xi_{l_4} \\
\mathcal{G}_1^1 = & P_{x_2}q_1S_1^3 + 3A_1P_{x_2}q_1S_1^5 - P_{x_2}q_2S_2^3 - 3A_2P_{x_2}q_2S_2^5 - 3A_1P_{x_2}q_1S_1^5x_1 \\
& - (3/2)A_1P_{x_2}q_1S_1^5\eta_{l_4} + P_{x_2}q_2S_2^3x_1 + 3A_2P_{x_2}q_2S_2^5x_1 - 3A_1P_{x_2}q_1S_1^5\xi_{l_4} + P_{x_2}q_2S_2^3\eta_{l_4} \\
& - P_{x_2}q_1S_1^3\xi_{l_4} + (3/2)A_2P_{x_2}q_2S_2^5x_2 + P_{x_2}q_2S_2^3\xi_{l_4} - P_{x_2}q_1S_1^3\eta_{l_4} - (3/2)A_1P_{x_2}q_1S_1^5x_2 \\
& - P_{x_2}q_1S_1^3x_1 + P_{x_2}q_2S_2^3x_2 - P_{x_2}q_1S_1^3x_2 \\
\mathcal{G}_2^1 = & -P_{x_2}q_1S_1^3 - 3A_1P_{x_2}q_1S_1^5P_{x_2}q_2S_2^3 + 3A_2P_{x_2}q_2S_2^5.
\end{aligned}$$

5.2. The second order solution

Setting $n = 2$ in Eq. (26) we obtain the required coefficients in Eq. (27) to yield the second order solution as;

$$\begin{aligned} \frac{\partial^2 \mathcal{H}}{\partial x_1^2} &= n^2 - q_1(1-\mu)S_1^3 + 3q_1(1-\mu)(x_1 + \xi_{l_1} + \mu)^2 S_1^5 - (3/2)q_1 A_1(1-\mu)S_1^5 \\ &+ (15/2)q_1 A_1(1-\mu)(x_1 + \xi_{l_1} + \mu)^2 S_1^7 - q_2 \mu S_2^3 + 3q_2 \mu(x_1 + \xi_{l_1} + \mu - 1)^2 S_2^5 \\ &- (3/2)q_2 A_2 \mu S_2^5 + (15/2)q_2 A_2 \mu(x_1 + \xi_{l_1} + \mu - 1)^2 S_2^7, \end{aligned} \quad (30.1)$$

$$\begin{aligned} \frac{\partial^2 \mathcal{H}}{\partial x_2^2} &= n^2 - q_1(1-\mu)S_1^3 + 3q_1(1-\mu)(x_2 + \eta_{l_1})^2 S_1^5 - (3/2)q_1 A_1(1-\mu)S_1^5 \\ &+ (15/2)q_1 A_1(1-\mu)(x_2 + \eta_{l_1})^2 S_1^7 - q_2 \mu S_2^3 + 3q_2 \mu(x_2 + \eta_{l_1})^2 S_2^5 \\ &- (3/2)q_2 A_2 \mu S_2^5 + (15/2)q_2 A_2 \mu(x_2 + \eta_{l_1})^2 S_2^7, \end{aligned} \quad (30.2)$$

$$\begin{aligned} \frac{\partial^2 \mathcal{H}}{\partial x_1 \partial x_2} &= 3q_1(1-\mu)(x_1 + \xi_{l_1} + \mu)(x_2 + \eta_{l_1})S_1^5 + (15/2)q_1 A_1(1-\mu)(x_1 + \xi_{l_1} + \mu) \\ &\times (x_2 + \eta_{l_1})S_1^7 + 3q_2 \mu(x_1 + \xi_{l_1} + \mu - 1)(x_2 + \eta_{l_1})S_2^5 \\ &+ (15/2)q_2 A_2 \mu(x_1 + \xi_{l_1} + \mu - 1)(x_2 + \eta_{l_1})S_2^7, \end{aligned} \quad (30.3)$$

$$\frac{\partial^2 \mathcal{H}}{\partial P_{x_1} \partial P_{x_1}} = 1, \quad (30.4)$$

$$\frac{\partial^2 \mathcal{H}}{\partial P_{x_2} \partial P_{x_2}} = 1, \quad (30.5)$$

$$\frac{\partial^2 \mathcal{H}}{\partial x_1 \partial P_{x_2}} = -n. \quad (30.6)$$

Substituting Eqs. (30.1)–(30.6) into Eq. (26) and neglecting the very small magnitude terms yields the second order solution as

$$\left. \begin{aligned} D^{(2)}x_1 &= \sum_{i=0}^2 \mathcal{J}_{i,2} \mu^i, \\ D^{(2)}x_2 &= \sum_{i=0}^2 \mathcal{N}_i^2 \mu^i, \\ D^{(2)}P_{x_1} &= \sum_{i=0}^2 \mathcal{K}_i^2 \mu^i, \\ D^{(2)}P_{x_2} &= \sum_{i=0}^2 \mathcal{G}_i^2 \mu^i. \end{aligned} \right\} \quad (31)$$

Where the nonvanishing included coefficients are given by

$$\begin{aligned}
\mathcal{J}_0^2 = & 2P_{x_1} - 2nP_{x_1}x_1 - 2n^2P_{x_1}x_1^2 - 2n^2x_1x_2 - 2n^3x_1x_2^2 + 2nP_{x_2}x_1 - 2n^2P_{x_2}x_1^2 \\
& + P_{x_1}q_1S_1^3x_1 + P_{x_2}q_1S_1^3x_1 + (3/2)A_1P_{x_1}q_1S_1^5x_1 + (3/2)A_1P_{x_2}q_1S_1^5x_1 + q_1S_1^3x_1x_2 \\
& + 2nq_1S_1^3x_1x_2 + (3/2)A_1q_1S_1^5x_1x_2 + 3nA_1q_1S_1^5x_1x_2 + q_1S_1^3x_1\eta_{L_1} \\
& + nq_1S_1^3x_1\eta_{L_1} + (3/2)A_1q_1S_1^5x_1\eta_{L_1} + (3/2)nA_1q_1S_1^5x_1\eta_{L_1} - 3P_{x_1}q_1S_1^5x_1^2\eta_{L_1} \\
& - 3P_{x_2}q_1S_1^5x_1^2\eta_{L_1} + 3nq_1S_1^5x_1^3\eta_{L_1} - 6P_{x_2}q_1S_1^5x_1x_2\eta_{L_1} + 3nq_1S_1^5x_1^2x_2\eta_{L_1} \\
& - 3P_{x_2}q_1S_1^5x_1\eta_{L_1}^2 - (15/2)A_1P_{x_2}q_1S_1^7x_1\eta_{L_1}^2 + 3nq_1S_1^5x_1^2\eta_{L_1}^2 + (15/2)nA_1q_1S_1^7x_1^2\eta_{L_1}^2 \\
& + q_1S_1^3x_1\xi_{L_1} - nq_1S_1^3x_1\xi_{L_1} + (3/2)A_1q_1S_1^5x_1\xi_{L_1} - (3/2)nA_1q_1S_1^5x_1\xi_{L_1} \\
& - 6P_{x_1}q_1S_1^5x_1^2\xi_{L_1} - 3P_{x_1}q_1S_1^5x_1x_2\xi_{L_1} - 3P_{x_2}q_1S_1^5x_1x_2\xi_{L_1} - 3nq_1S_1^5x_1^2x_2\xi_{L_1} \\
& - 3nq_1S_1^5x_1x_2^2\xi_{L_1} - 3P_{x_1}q_1S_1^5x_1\eta_{L_1}\xi_{L_1} - 3P_{x_2}q_1S_1^5x_1\eta_{L_1}\xi_{L_1} - (15/2)A_1P_{x_1}q_1S_1^7x_1\eta_{L_1}\xi_{L_1} \\
& - (15/2)A_1P_{x_2}q_1S_1^7x_1\eta_{L_1}\xi_{L_1} + 3nq_1S_1^5x_1^2\eta_{L_1}\xi_{L_1} + (15/2)nA_1q_1S_1^7x_1^2\eta_{L_1}\xi_{L_1} \\
& - 3P_{x_1}q_1S_1^5x_1\xi_{L_1}^2 - 3nq_1S_1^5x_1x_2\eta_{L_1}\xi_{L_1} - 3nq_1S_1^5x_1x_2\xi_{L_1}^2 \\
& - (15/2)nA_1q_1S_1^7x_1x_2\xi_{L_1}^2 - (15/2)A_1P_{x_1}q_1S_1^7x_1\xi_{L_1}^2 - (15/2)nA_1q_1S_1^7x_1x_2\eta_{L_1}\xi_{L_1} \\
\mathcal{J}_1^2 = & q_1S_1^3x_1 - nq_1S_1^3x_1 - P_{x_1}q_1S_1^3x_1 - P_{x_2}q_1S_1^3x_1 + (3/2)A_1q_1S_1^5x_1 - (3/2)nA_1q_1S_1^5x_1 \\
& - q_2S_2^3x_1 + q_2S_2^3x_1^2 - q_1S_1^3x_1x_2 - 2nq_1S_1^3x_1x_2 + q_2S_2^3x_1x_2 + 2nq_2S_2^3x_1x_2 \\
& - 3nq_2S_2^5x_1x_2 - nq_1S_1^3x_1\eta_{L_1} - q_1S_1^3x_1\eta_{L_1} - (3/2)nA_1q_1S_1^5x_1\eta_{L_1} - (3/2)A_1q_1S_1^5x_1\eta_{L_1} \\
& - 3P_{x_1}q_1S_1^5x_1\eta_{L_1} - 3P_{x_2}q_1S_1^5x_1\eta_{L_1} + q_2S_2^3x_1\eta_{L_1} + (3/2)A_2q_2S_2^5x_1\eta_{L_1} \\
& + (3/2)nA_2q_2S_2^5x_1\eta_{L_1} + 3P_{x_1}q_2S_2^5x_1\eta_{L_1} - 2nq_2S_2^3x_1^2 + q_2S_2^3x_1^2 - q_1S_1^3x_1x_2 \\
& - 2nq_1S_1^3x_1x_2 + q_2S_2^3x_1x_2 - nq_1S_1^3x_1\eta_{L_1} + nq_2S_2^3x_1\eta_{L_1} + 3P_{x_2}q_2S_2^5x_1\eta_{L_1} \\
& + 3nq_1S_1^5x_1^2\eta_{L_1} - 3nq_2S_2^5x_1^2\eta_{L_1} + 3nq_2S_2^5x_1x_2\eta_{L_1} + 3P_{x_2}q_1S_1^5x_1\eta_{L_1}^2 \\
& - 3P_{x_2}q_2S_2^5x_1\eta_{L_1}^2 - 3nq_1S_1^5x_1^2\eta_{L_1}^2 + (3/2)nA_1q_1S_1^5x_1\xi_{L_1} - (3/2)A_1q_1S_1^5x_1\xi_{L_1} \\
& - 6P_{x_1}q_1S_1^5x_1\xi_{L_1} - nq_2S_2^3x_1\xi_{L_1} + (3/2)A_2q_2S_2^5x_1\xi_{L_1} - (3/2)nA_2q_2S_2^5x_1\xi_{L_1} \\
& + 6P_{x_1}q_2S_2^5x_1\xi_{L_1} - 6nq_1S_1^5x_1x_2\xi_{L_1} + q_2S_2^3x_1\xi_{L_1} + 6nq_2S_2^5x_1x_2\xi_{L_1} - 3nq_2S_2^5x_1x_2\xi_{L_1}^2
\end{aligned}$$

$$\begin{aligned}
& + 3nq_2S_2^5x_1^2\eta_{L_i}\xi_{L_i} + 3P_{x_2}q_1S_1^5x_1\eta_{L_i}\xi_{L_i} + 3nq_1S_1^5x_1x_2\eta_{L_i}\xi_{L_i} - 3nq_2S_2^5x_1x_2\eta_{L_i}\xi_{L_i} \\
& - 3nq_1S_1^5x_1^2\eta_{L_i}\xi_{L_i} + 3P_{x_1}q_1S_1^5x_1^2\xi_{L_i}^2 - 3P_{x_1}q_2S_2^5x_1^2\xi_{L_i}^2 + 3nq_1S_1^5x_1x_2\xi_{L_i}^2 \\
\mathcal{J}_2^2 = & -\mu^2q_1S_1^3x_1 + n\mu^2q_1S_1^3x_1 - q_1S_1^3x_1 + nq_1S_1^3x_1 \\
\mathcal{N}_0^2 = & 2P_{x_2} - 2nx_1 - 2nP_{x_1}x_2 - 2n^2P_{x_1}x_2 + 2nP_{x_2}x_2 - 2n^2P_{x_2}x_2 + q_1S_1^3x_1x_2 \\
& + P_{x_1}q_1S_1^3x_2 + (3/2)A_1P_{x_1}q_1S_1^5x_2 + (3/2)A_1P_{x_2}q_1S_1^5x_2 - 2n^2x_1x_2 + 2n^3x_1x_2 \\
& - 2nq_1S_1^3x_1x_2 + (3/2)A_1q_1S_1^5x_1x_2 - 3nA_1q_1S_1^5x_1x_2 + q_1S_1^3x_2\eta_{L_i} + nq_1S_1^3x_2\eta_{L_i} \\
& - 2n^3x_2^2 + q_1S_1^3x_2^2 + 2nq_1S_1^3x_2^2 - 2n^2x_2^2 + (3/2)A_1q_1S_1^5x_2^2 + 3nA_1q_1S_1^5x_2^2 \\
& + (3/2)A_1q_1S_1^5x_2\eta_{L_i} + (3/2)nA_1q_1S_1^5x_2\eta_{L_i} + n\mu q_2S_2^3x_2\eta_{L_i} - 3P_{x_1}q_1S_1^5x_1x_2\eta_{L_i} \\
& - 3P_{x_2}q_1S_1^5x_1x_2\eta_{L_i} - 6P_{x_2}q_1S_1^5x_2^2\eta_{L_i} + (15/2)nA_1q_1S_1^7x_1x_2\eta_{L_i}^2 + 3nq_1S_1^5x_1^2x_2\eta_{L_i} \\
& + 3nq_1S_1^5x_1x_2\eta_{L_i}^2 - (15/2)A_1P_{x_2}q_1S_1^7x_2\eta_{L_i}^2 + 3nq_1S_1^5x_1x_2\eta_{L_i} - 3P_{x_2}q_1S_1^5x_2\eta_{L_i}^2 \\
& + (3/2)A_1q_1S_1^5x_2\xi_{L_i} + q_1S_1^3x_2\xi_{L_i} - nq_1S_1^3x_2\xi_{L_i} - (3/2)nA_1q_1S_1^5x_2\xi_{L_i} + \mu q_2S_2^3x_2\xi_{L_i} \\
& - 6P_{x_1}q_1S_1^5x_1x_2\xi_{L_i} - 3P_{x_1}q_1S_1^5x_2^2\xi_{L_i} - 3P_{x_2}q_1S_1^5x_2^2\xi_{L_i} - (15/2)A_1P_{x_2}q_1S_1^7x_2\eta_{L_i}\xi_{L_i} \\
& - 3nq_1S_1^5x_1x_2^2\xi_{L_i} - 3nq_1S_1^5x_2^2\xi_{L_i} - 3P_{x_1}q_1S_1^5x_2\eta_{L_i}\xi_{L_i} + (15/2)nA_1q_1S_1^7x_1x_2\eta_{L_i}\xi_{L_i} \\
& - (15/2)nA_1q_1S_1^7x_2^2\eta_{L_i}\xi_{L_i} - 3P_{x_2}q_1S_1^5x_2\eta_{L_i}\xi_{L_i} - 3nq_1S_1^5x_2^2\eta_{L_i}\xi_{L_i} - 3P_{x_1}q_1S_1^5x_2^2\xi_{L_i}^2 \\
& - (15/2)A_1P_{x_1}q_1S_1^7x_2\eta_{L_i}\xi_{L_i} - 3nq_1S_1^5x_2^2\xi_{L_i}^2 + 3nq_1S_1^5x_1x_2\eta_{L_i}\xi_{L_i} - \mu q_1S_1^3x_2\xi_{L_i} \\
& + P_{x_2}q_1S_1^3x_2 - (15/2)A_1P_{x_1}q_1S_1^7x_2\xi_{L_i}^2 - (15/2)nA_1q_1S_1^7x_2^2\xi_{L_i}^2 \\
\mathcal{N}_1^2 = & q_1S_1^3x_2 - nq_1S_1^3x_2 - P_{x_1}q_1S_1^3x_2 - P_{x_2}q_1S_1^3x_2 + (3/2)A_1q_1S_1^5x_2 - (3/2)nA_1q_1S_1^5x_2 \\
& - q_2S_2^3x_2 + nq_2S_2^3x_2 + P_{x_1}q_2S_2^3x_2 + P_{x_2}q_2S_2^3x_2 - (3/2)A_2q_2S_2^5x_2 + (3/2)nA_2q_2S_2^5x_2 \\
& - q_1S_1^3x_2^2 - 2nq_1S_1^3x_2^2 - (3/2)A_1q_1S_1^5x_1x_2 + q_2S_2^3x_1x_2 - 3P_{x_1}q_2S_2^5x_2 - q_1S_1^3x_1x_2 \\
& + 2nq_1S_1^3x_1x_2 - 2nq_2S_2^3x_1x_2 + q_2S_2^3x_2^2 + 2nq_2S_2^3x_2^2 - 3nq_2S_2^5x_2^2 - q_1S_1^3x_2\eta_{L_i} \\
& - nq_1S_1^3x_2\eta_{L_i} - (3/2)A_1q_1S_1^5x_2\eta_{L_i} - (3/2)nA_1q_1S_1^5x_2\eta_{L_i} - 3P_{x_1}q_1S_1^5x_2\eta_{L_i} \\
& - 3P_{x_2}q_1S_1^5x_2\eta_{L_i} + (3/2)A_2q_2S_2^5x_2\eta_{L_i} + (3/2)nA_2q_2S_2^5x_2\eta_{L_i} + q_2S_2^3x_2\eta_{L_i}
\end{aligned}$$

$$\begin{aligned}
& + 3P_{x_1} q_2 S_2^5 x_2 \eta_{L_i} + 3P_{x_2} q_2 S_2^5 x_2 \eta_{L_i} + 3nq_1 S_1^5 x_1 x_2 \eta_{L_i} - 3nq_2 S_2^5 x_1 x_2 \eta_{L_i} \\
& + 3nq_2 S_2^5 x_2^2 \eta_{L_i} + 3P_{x_2} q_1 S_1^5 x_2 \eta_{L_i}^2 - 3P_{x_2} q_2 S_2^5 x_2 \eta_{L_i}^2 - 3nq_1 S_1^5 x_1 x_2 \eta_{L_i}^2 \\
& + nq_1 S_1^3 x_2 \xi_{L_i} - (3/2)A_1 q_1 S_1^5 x_2 \xi_{L_i} + (3/2)nA_1 q_1 S_1^5 x_2 \xi_{L_i} - 6P_{x_1} q_1 S_1^5 x_2 \xi_{L_i} \\
& - nq_2 S_2^3 x_2 \xi_{L_i} + (3/2)A_2 q_2 S_2^5 x_2 \xi_{L_i} - (3/2)nA_2 q_2 S_2^5 x_2 \xi_{L_i} - 6nq_1 S_1^5 x_2^2 \xi_{L_i} \\
& + 6nq_2 S_2^5 x_2^2 \xi_{L_i} + 3P_{x_1} q_1 S_1^5 x_2 \eta_{L_i} \xi_{L_i} + 6P_{x_1} q_2 S_2^5 x_2 \xi_{L_i} + 3P_{x_2} q_1 S_1^5 x_2 \eta_{L_i} \xi_{L_i} \\
& - 3P_{x_1} q_2 S_2^5 x_2 \eta_{L_i} \xi_{L_i} - 3P_{x_2} q_2 S_2^5 x_2 \eta_{L_i} \xi_{L_i} + 3nq_1 S_1^5 x_2^2 \xi_{L_i}^2 - 3P_{x_1} q_2 S_2^5 x_2 \xi_{L_i}^2 \\
& + 3nq_1 S_1^5 x_2^2 \eta_{L_i} \xi_{L_i} + 3P_{x_1} q_1 S_1^5 x_2 \xi_{L_i}^2 - 3nq_1 S_1^5 x_1 x_2 \eta_{L_i} \xi_{L_i} - 3nq_2 S_2^5 x_2^2 \xi_{L_i}^2 \\
& - 3nq_2 S_2^5 x_2^2 \eta_{L_i} \xi_{L_i} + 3nq_2 S_2^5 x_1 x_2 \eta_{L_i} \xi_{L_i} - 3nq_1 S_1^5 x_2^2 \eta_{L_i} + 3nq_2 S_2^5 x_1 x_2 \eta_{L_i}^2
\end{aligned}$$

$$N_2^2 = \mu^2 q_2 S_2^3 x_2 - n \mu^2 q_2 S_2^3 x_2 - \mu^2 q_1 S_1^3 x_2 + n \mu^2 q_1 S_1^3 x_2$$

$$\begin{aligned}
\mathcal{K}_0^2 & = 2nP_{x_1} P_{x_2} - 2n^2 P_{x_1} P_{x_2} + P_{x_1}^2 q_1 S_1^3 + P_{x_1} P_{x_2} q_1 S_1^3 - 3P_{x_1}^2 q_1 S_1^5 x_2 \xi_{L_i} - 2nP_{x_1} q_1 S_1^3 x_1 \\
& + (3/2)A_1 P_{x_1} P_{x_2} q_1 S_1^5 - 2n^2 x_1 - 2n^2 P_{x_1} x_1 + 2n^3 P_{x_1} x_1 + 2q_1 S_1^3 x_1 + P_{x_1} q_1 S_1^3 x_1 + 2nP_{x_2} \\
& + (3/2)A_1 P_{x_1}^2 q_1 S_1^5 + 3A_1 q_1 S_1^5 x_1 + (3/2)A_1 P_{x_1} q_1 S_1^5 x_1 - 3nA_1 P_{x_1} q_1 S_1^5 x_1 - 2n^2 P_{x_1} x_2 \\
& - 2n^3 P_{x_1} x_2 + P_{x_1} q_1 S_1^3 x_2 + 2nP_{x_1} q_1 S_1^3 x_2 + (3/2)A_1 P_{x_1} q_1 S_1^5 x_2 + 3nA_1 P_{x_1} q_1 S_1^5 x_2 \\
& + (15/2)nA_1 P_{x_1} q_1 S_1^7 x_1 \eta_{L_i}^2 + nP_{x_1} q_1 S_1^3 \eta_{L_i} + (3/2)A_1 P_{x_1} q_1 S_1^5 \eta_{L_i} + (3/2)nA_1 P_{x_1} q_1 S_1^5 \eta_{L_i} \\
& - 3P_{x_1}^2 q_1 S_1^5 x_1 \eta_{L_i} - (15/2)A_1 P_{x_1} P_{x_2} q_1 S_1^7 \eta_{L_i}^2 + P_{x_1} q_1 S_1^3 \xi_{L_i} - nP_{x_1} q_1 S_1^5 \xi_{L_i} - 3P_{x_1} P_{x_2} q_1 S_1^5 \eta_{L_i}^2 \\
& + 3A_1 q_1 S_1^5 \xi_{L_i} + (3/2)A_1 P_{x_1} q_1 S_1^5 \xi_{L_i} - (3/2)nA_1 P_{x_1} q_1 S_1^5 \xi_{L_i} - 6P_{x_1}^2 q_1 S_1^5 x_1 \xi_{L_i} - 2n^2 P_{x_1}^2 \\
& - 3P_{x_1}^2 q_1 S_1^5 \xi_{L_i}^2 - 3nP_{x_1} q_1 S_1^5 x_1 x_2 \xi_{L_i} + 2q_1 S_1^3 \xi_{L_i} - 3P_{x_1}^2 q_1 S_1^5 \eta_{L_i} \xi_{L_i} - 3P_{x_1} P_{x_2} q_1 S_1^5 \eta_{L_i} \xi_{L_i} \\
& - (15/2)A_1 P_{x_1}^2 q_1 S_1^7 \eta_{L_i} \xi_{L_i} - (15/2)A_1 P_{x_1} P_{x_2} q_1 S_1^7 \eta_{L_i} \xi_{L_i} + (15/2)nA_1 P_{x_1} q_1 S_1^7 x_1 \eta_{L_i} \xi_{L_i} \\
& - 3nP_{x_1} q_1 S_1^5 x_2 \eta_{L_i} \xi_{L_i} - (15/2)A_1 P_{x_1}^2 q_1 S_1^7 \xi_{L_i}^2 + P_{x_1} q_1 S_1^3 \eta_{L_i} - 3nP_{x_1} q_1 S_1^5 x_2^2 \xi_{L_i} - 2nP_{x_1}^2 \\
& - 3nP_{x_1} q_1 S_1^5 x_2 \xi_{L_i}^2 - (15/2)nA_1 P_{x_1} q_1 S_1^7 x_2 \xi_{L_i}^2 - 3P_{x_1} P_{x_2} q_1 S_1^5 x_2 \xi_{L_i} + 3P_{x_1} q_1 S_1^3 \xi_{L_i}^3 \\
& - 2nP_{x_1} q_1 S_1^3 \xi_{L_i}^3 + 3\mu q_2 S_2^3 \xi_{L_i}^3 + 3nP_{x_1} q_1 S_1^5 x_1 \eta_{L_i} \xi_{L_i} - (15/2)nA_1 P_{x_1} q_1 S_1^7 x_2 \eta_{L_i} \xi_{L_i} \\
\mathcal{K}_1^2 & = -P_{x_1} q_1 S_1^3 \xi_{L_i} + nP_{x_1} q_1 S_1^3 \xi_{L_i} - 3A_1 q_1 S_1^5 \xi_{L_i} - (3/2)A_1 P_{x_1} q_1 S_1^5 \xi_{L_i} - nP_{x_1} q_1 S_1^3 \eta_{L_i}
\end{aligned}$$

$$\begin{aligned}
& + (3/2)nA_1P_{x_1}q_1S_1^5\xi_{L_i} + 3A_2q_2S_2^5x_1 - nP_{x_1}q_2S_2^3\xi_{L_i} + 3A_2q_2S_2^5\xi_{L_i} + 3nP_{x_1}q_2S_2^5x_1\eta_{L_i}^2 \\
& - (3/2)nA_2P_{x_1}q_2S_2^5\xi_{L_i} + 2q_2S_2^3\xi_{L_i} + 6P_{x_1}^2q_2S_2^5\xi_{L_i} - 6nP_{x_1}q_1S_1^5x_2\xi_{L_i} + 6nP_{x_1}q_2S_2^5x_2\xi_{L_i} \\
& + 3P_{x_1}^2q_1S_1^5\eta_{L_i}\xi_{L_i} + 3P_{x_1}P_{x_2}q_1S_1^5\eta_{L_i}\xi_{L_i} + 3nP_{x_1}q_1S_1^5x_2\eta_{L_i}\xi_{L_i} + 3P_{x_1}^2q_1S_1^5\xi_{L_i}^2 + nP_{x_1}q_2S_2^3 \\
& - (3/2)nA_1P_{x_1}q_1S_1^5\eta_{L_i} + (3/2)A_2P_{x_1}q_2S_2^5\xi_{L_i} - 3P_{x_1}^2q_2S_2^5\xi_{L_i}^2 + 3nP_{x_1}q_1S_1^5x_2\xi_{L_i}^2 \\
& - 3nP_{x_1}q_2S_2^5x_2\xi_{L_i}^2 - 3nP_{x_1}q_2S_2^5x_2\eta_{L_i}\xi_{L_i} - 3q_1S_1^3\xi_{L_i}^3 - 3P_{x_1}^2q_2S_2^5\eta_{L_i}\xi_{L_i} - P_{x_1}q_2S_2^3 \\
& - 3P_{x_1}P_{x_2}q_2S_2^5\eta_{L_i}\xi_{L_i} - 3nP_{x_1}q_1S_1^5x_1\eta_{L_i}\xi_{L_i} + 3nP_{x_1}q_2S_2^5x_1\eta_{L_i}\xi_{L_i} - (3/2)nA_1P_{x_1}q_1S_1^5 \\
& + 2q_1S_1^3 + P_{x_1}q_1S_1^3 - nP_{x_1}q_1S_1^3 - P_{x_1}^2q_1S_1^3 - P_{x_1}P_{x_2}q_1S_1^3 + 3A_1q_1S_1^5 + (3/2)A_1P_{x_1}q_1S_1^5 \\
& - 2q_2S_2^3 + P_{x_1}P_{x_2}q_2S_2^3 - 3A_2q_2S_2^5 - (3/2)A_2P_{x_1}q_2S_2^5 + (3/2)nA_2P_{x_1}q_2S_2^5 + P_{x_1}^2q_2S_2^3 \\
& - 2q_1S_1^3x_1 - P_{x_1}q_1S_1^3x_1 + 2nP_{x_1}q_1S_1^3x_1 - 3A_1q_1S_1^5x_1 + 2q_2S_2^3x_1 + P_{x_1}q_2S_2^3x_1 \\
& - 2nP_{x_1}q_1S_1^3x_2 + P_{x_1}q_2S_2^3x_2 - 2nP_{x_1}q_2S_2^3x_1 - 3P_{x_1}^2q_2S_2^5 - P_{x_1}q_1S_1^3x_2 + P_{x_1}q_2S_2^3\xi_{L_i} \\
& - (3/2)A_1P_{x_1}q_1S_1^5\eta_{L_i} + 2nP_{x_1}q_2S_2^3x_2 - 3nP_{x_1}q_2S_2^5x_2 - P_{x_1}q_1S_1^3\eta_{L_i} - 3P_{x_1}^2q_1S_1^5\eta_{L_i} \\
& + (3/2)A_2P_{x_1}q_2S_2^5\eta_{L_i} - 3P_{x_1}P_{x_2}q_1S_1^5\eta_{L_i} - 3P_{x_1}P_{x_2}q_2S_2^5\eta_{L_i}^2 + nP_{x_1}q_2S_2^3\eta_{L_i} \\
& + 3P_{x_1}^2q_2S_2^5\eta_{L_i} - 3P_{x_1}P_{x_2}q_1S_1^5x_1\eta_{L_i} - 3nP_{x_1}q_2S_2^5\eta_{L_i} + (3/2)nA_2P_{x_1}q_2S_2^5\eta_{L_i}^2 \\
& + 3nP_{x_1}q_1S_1^5\eta_{L_i} - 3nP_{x_1}q_1S_1^5x_2\eta_{L_i} - 6P_{x_1}P_{x_2}q_1S_1^3x_2\eta_{L_i} + 3nP_{x_1}q_2S_2^3x_2\eta_{L_i} \\
& + 3nP_{x_1}q_1S_1^5x_1x_2\eta_{L_i} + 3P_{x_1}^2P_{x_2}q_1S_1^5\eta_{L_i}^2 + 3P_{x_1}P_{x_2}q_2S_2^5\eta_{L_i} + 3nP_{x_1}q_1S_1^5x_1\eta_{L_i} \\
& + 3nP_{x_1}q_1S_1^7\eta_{L_i}^2 - 3nP_{x_1}q_1S_1^5x_1\eta_{L_i}^2 - 2q_1S_1^3\xi_{L_i} - 6P_{x_1}^2q_1S_1^5\xi_{L_i} + P_{x_1}q_2S_2^3\eta_{L_i} \\
\mathcal{K}_2^2 = & -2q_1S_1^3 - \mu^2P_{x_1}q_1S_1^3 + nP_{x_1}q_1S_1^3 + 2q_2S_2^3 + P_{x_1}q_2S_2^3 - nP_{x_1}q_2S_2^3 - 3A_1q_1S_1^5 + 3A_2q_2S_2^5 \\
\mathcal{G}_0^2 = & -2nP_{x_1}P_{x_2} - 2n^2P_{x_1}P_{x_2} + 2nP_{x_2}^2 - 2n^2P_{x_2}^2 + P_{x_1}P_{x_2}q_1S_1^3 + P_{x_2}^2q_1S_1^3 + P_{x_2}q_1S_1^3x_1 \\
& + (3/2)A_1P_{x_1}P_{x_2}q_1S_1^5 + (3/2)A_1P_{x_2}^2q_1S_1^5 - 2n^2P_{x_2}x_1 + 2n^3P_{x_2}x_1 - 2nP_{x_2}q_1S_1^3x_1 \\
& + (3/2)A_1P_{x_2}q_1S_1^5x_1 - 3nA_1P_{x_2}q_1S_1^5x_1 - 2n^2x_2 - 2n^2P_{x_2}x_2 - 2n^3P_{x_2}x_2 + 2q_1S_1^3x_2 \\
& + P_{x_2}q_1S_1^3x_2 + 2nP_{x_2}q_1S_1^3x_2 + 3A_1q_1S_1^5x_2 + P_{x_2}q_1S_1^3\eta_{L_i} + (3/2)A_1P_{x_2}q_1S_1^5x_2 \\
& + 3nA_1P_{x_2}q_1S_1^5x_2 + 3A_1q_1S_1^5\eta_{L_i} + 2q_1S_1^3\eta_{L_i} + nP_{x_2}q_1S_1^3\eta_{L_i} + (3/2)nA_1P_{x_2}q_1S_1^5\eta_{L_i}
\end{aligned}$$

$$\begin{aligned}
& + (3/2)A_1P_{x_2}q_1S_1^5\eta_{l_i} - 3P_{x_1}P_{x_2}q_1S_1^5x_1\eta_{l_i} + 3nP_{x_2}q_1S_1^5x_1^2\eta_{l_i} - 6P_{x_2}^2q_1S_1^5x_2\eta_{l_i} \\
& - 3P_{x_1}P_{x_2}q_1S_1^5x_2\xi_{l_i} - 3P_{x_2}^2q_1S_1^5x_2\xi_{l_i} - 3nP_{x_2}q_1S_1^5x_2^2\xi_{l_i} - 3P_{x_1}P_{x_2}q_1S_1^5\eta_{l_i}\xi_{l_i} - 2nP_{x_1} \\
& - 3P_{x_1}P_{x_2}q_1S_1^5\xi_{l_i}^2 + 3\mu P_{x_1}P_{x_2}q_1S_1^5\xi_{l_i}^2 - (15/2)A_1P_{x_2}^2q_1S_1^7\eta_{l_i}\xi_{l_i} - 3nP_{x_2}q_1S_1^5x_2\xi_{l_i}^2 \\
& + (15/2)nA_1P_{x_2}q_1S_1^7x_1\eta_{l_i}\xi_{l_i} + 3nP_{x_2}q_1S_1^5x_1\eta_{l_i}\xi_{l_i} - (15/2)A_1P_{x_1}P_{x_2}q_1S_1^7\xi_{l_i}^2 + P_{x_2}q_1S_1^3\xi_{l_i} \\
& - 3nP_{x_2}q_1S_1^5x_2\eta_{l_i}\xi_{l_i} + 3nP_{x_2}q_1S_1^5x_1x_2\eta_{l_i} - 3P_{x_2}^2q_1S_1^5\eta_{l_i}^2 - (15/2)A_1P_{x_2}^2q_1S_1^7\eta_{l_i}^2 \\
& + (15/2)nA_1P_{x_2}q_1S_1^7x_1\eta_{l_i}^2 + 3nP_{x_2}q_1S_1^5x_1\eta_{l_i}^2 + (3/2)A_1P_{x_2}q_1S_1^5\xi_{l_i} - nP_{x_2}q_1S_1^3\xi_{l_i} \\
& - 6P_{x_1}P_{x_2}q_1S_1^5x_1\xi_{l_i} - (3/2)nA_1P_{x_2}q_1S_1^5\xi_{l_i} - 3nP_{x_2}q_1S_1^5x_1x_2\xi_{l_i} - 3P_{x_2}^2q_1S_1^5\eta_{l_i}\xi_{l_i} \\
\mathcal{G}_1^2 = & (3/2)A_1P_{x_2}q_1S_1^5 - (3/2)nA_1P_{x_2}q_1S_1^5 - P_{x_2}^2q_1S_1^3\mu P_{x_2}q_1S_1^3 - nP_{x_2}q_1S_1^3 - P_{x_1}P_{x_2}q_1S_1^3 \\
& - P_{x_2}q_2S_2^3 + nP_{x_2}q_2S_2^3 + P_{x_1}P_{x_2}q_2S_2^3 + P_{x_2}^2q_2S_2^3 - (3/2)A_2P_{x_2}q_2S_2^5 + (3/2)nA_2P_{x_2}q_2S_2^5 \\
& - 3P_{x_1}P_{x_2}q_2S_2^5 - P_{x_2}q_1S_1^3x_1 + 2nP_{x_2}q_1S_1^3x_1 + P_{x_2}q_2S_2^3x_1 - 2nP_{x_2}q_2S_2^3x_1 - 2q_1S_1^3x_2 \\
& - P_{x_2}q_1S_1^3x_2 - 2nP_{x_2}q_1S_1^3x_2 - 3A_1q_1S_1^5x_2 + 2q_2S_2^3x_2 + P_{x_2}q_2S_2^3x_2 + 2nP_{x_2}q_2S_2^3x_2 \\
& - P_{x_2}q_1S_1^3\eta_{l_i} + 2q_2S_2^3\eta_{l_i} + 3A_2q_2S_2^5x_2 - 3nP_{x_2}q_2S_2^5x_2 - 2q_1S_1^3\eta_{l_i} - nP_{x_2}q_1S_1^3\eta_{l_i} \\
& - 3A_1q_1S_1^5\eta_{l_i} - (3/2)A_1P_{x_2}q_1S_1^5\eta_{l_i} - (3/2)nA_1P_{x_2}q_1S_1^5\eta_{l_i} - 3P_{x_1}P_{x_2}q_1S_1^5\eta_{l_i} \\
& + P_{x_2}q_2S_2^3\eta_{l_i} + nP_{x_2}q_2S_2^3\eta_{l_i} + 3A_2q_2S_2^5\eta_{l_i} + (3/2)A_2P_{x_2}q_2S_2^5\eta_{l_i} + (3/2)nA_2P_{x_2}q_2S_2^5\eta_{l_i} \\
& + 3P_{x_1}P_{x_2}q_2S_2^5\eta_{l_i} + 3P_{x_2}^2q_2S_2^5\eta_{l_i} + 3nP_{x_2}q_1S_1^5x_1\eta_{l_i} - 3nP_{x_2}q_2S_2^5x_1\eta_{l_i} - 3nP_{x_2}q_1S_1^5x_2\eta_{l_i} \\
& + 3nP_{x_2}q_2S_2^5x_2\eta_{l_i} + 3P_{x_2}^2q_1S_1^5\eta_{l_i}^2 - 3P_{x_2}^2q_2S_2^5\eta_{l_i}^2 - 3n\mu P_{x_2}q_1S_1^5x_1\eta_{l_i}^2 - 6P_{x_1}P_{x_2}q_1S_1^5\xi_{l_i} \\
& + 3nP_{x_2}q_2S_2^5x_1\eta_{l_i}^2 + n\mu P_{x_2}q_1S_1^3\xi_{l_i} - (3/2)A_1P_{x_2}q_1S_1^5\xi_{l_i} + (3/2)nA_1P_{x_2}q_1S_1^5\xi_{l_i} + P_{x_2}q_2S_2^3\xi_{l_i} \\
& - nP_{x_2}q_2S_2^3\xi_{l_i} + (3/2)A_2P_{x_2}q_2S_2^5\xi_{l_i} - (3/2)nA_2P_{x_2}q_2S_2^5\xi_{l_i} + 6P_{x_1}P_{x_2}q_2S_2^5\xi_{l_i} - P_{x_2}q_1S_1^3\xi_{l_i} \\
& - 6nP_{x_2}q_1S_1^5x_2\xi_{l_i} + 3nP_{x_2}q_2S_2^5x_1\eta_{l_i}\xi_{l_i} + 3P_{x_1}P_{x_2}q_1S_1^5\eta_{l_i}\xi_{l_i} + 3P_{x_2}^2q_1S_1^5\eta_{l_i}\xi_{l_i} - 3P_{x_1}P_{x_2}q_2S_2^5\xi_{l_i}^2 \\
& - 3P_{x_2}^2q_2S_2^5\eta_{l_i}\xi_{l_i} - 3P_{x_1}P_{x_2}q_2S_2^5\eta_{l_i}\xi_{l_i} - 3nP_{x_2}q_2S_2^5x_2\eta_{l_i}\xi_{l_i} + 3nP_{x_2}q_1S_1^5x_2\xi_{l_i}^2 \\
& - 3nP_{x_2}q_2S_2^5x_2\xi_{l_i}^2 + 3nP_{x_2}q_1S_1^5x_2\eta_{l_i}\xi_{l_i} - 3nP_{x_2}q_1S_1^5x_1\eta_{l_i}\xi_{l_i} + 6nP_{x_2}q_2S_2^5x_2\xi_{l_i} - 3P_{x_2}^2q_1S_1^5\eta_{l_i} \\
\mathcal{G}_2^2 = & -P_{x_2}q_1S_1^3 + nP_{x_2}q_1S_1^3 + P_{x_2}q_2S_2^3 - nP_{x_2}q_2S_2^3
\end{aligned}$$

6. Conclusion

We can conclude our work in this research as follows: First we have outlined briefly the restricted three body problem, then we defined the infinitesimal orbits. We expressed the photogravitational oblate RTBP in both inertial and rotated coordinate systems. The Hamiltonian of the problem under investigation is constructed. Then it is transferred to any point of the equilibrium point as an origin. We have reviewed the Lie operator method, as a method of solution. Finally we have obtained the explicit first order as well as the second order solutions for the coordinates and their conjugate momenta of a test particle in an infinitesimal orbit around any equilibrium point.

Conflict of interests

The authors declare that there are no Conflict of interests associated with this work.

References

1. M. K. Ahmed, F. A. Abd El-Salam and S. E. Abd El-Bar, On the stability of the triangular Lagrangian equilibrium points in the relativistic restricted three-body problem, *Am. J. Appl. Sci.*, **3** (2006), 1993–1998.
2. C. N. Douskos, E. A. Perdios, On the stability of equilibrium points in the relativistic restricted three-body problem, *Celest. Mech. Dyn. Astron.*, **82** (2002), 317–321.
3. F. A. Abd El-Salam, S. E. Abd El-Bar, Formulation of the post-Newtonian equations of motion of the restricted three body problem, *Appl. Math.*, **2** (2011), 155–164.
4. F. A. Abd El-Salam, S. E. Abd El-Bar, Combined effects of oblateness and photogravitational perturbations of the stability of the equilibrium points in the relativistic RTBP, *Can. J. Phys.*, **93** (2015), 300–311.
5. F. A. Abd El-Salam, D. A. Katour, Relativistic restricted three body problem with oblateness and photo-gravitational corrections to triangular equilibrium points, *Astrophys. Space Sci.*, **351** (2014), 143–149.
6. F. A. Abd El-Salam, Stability of triangular equilibrium points in the elliptic restricted three body problem with oblate and triaxial primaries, *Astrophys. Space Sci.*, **357** (2015), 1–9.
7. R. Duncombe, V. Szebehely, *Methods of Astrodynamics and Celestial Mechanics*, New York: Academic press, 1966.
8. D. L. Richardson, Analytic construction of periodic orbits about the collinear points, *Celest. Mech. Dyn. Astron.*, **22** (1980), 241–253.
9. B. T. Barden, K. C. Howell, Fundamental motions near collinear libration points and their transitions, *J. Astronaut. Sci.*, **46** (1998), 361–378.
10. K. C. Howell, Families of orbits in the vicinity of the collinear libration points, *J. Astronaut. Sci.*, **49** (2001), 107–125.
11. G. Gómez, J. Masdemont and C. Simo, Quasihalo orbits associated with libration points, *J. Astronaut. Sci.*, **46** (1998), 135–176.
12. G. Gómez, A. Jorba, J. Masdemont, et al., Study of the transfer between halo orbits, *Acta Astronaut.*, **43** (1998), 493–520.

13. D. Selaru, C. Cucu-Dumitrescu, Infinitesimal orbit around Lagrange points in the elliptic restricted three body problem, *Celest. Mech. Dyn. Astron.*, **61** (1995), 333–346.
14. F. Namouni, C. D. Murray, The effect of eccentricity and inclination on the motion near the Lagrangian points L_4 and L_5 , *Celest. Mech. Dyn. Astron.*, **76** (2000), 131–138.
15. M. Corbera, J. Llibre, Periodic orbits of a collinear restricted three-body problem, *Celest. Mech. Dyn. Astron.*, **86** (2003), 163–183.
16. A. Hamdy, M. K. Ahmed, M. Radwan, et al., Infinitesimal orbits around the libration points in the elliptic restricted three body problem, *Earth, Moon, Planets*, **93** (2003), 261–273.
17. F. A. Abd El-Salam, Periodic and degenerate orbits around the equilibrium points in the relativistic restricted three-body problem, *Iran. J. Sci. Technol. A*, **43** (2018), 173–192.
18. A. H. Ibrahim, M. N. Ismail, A. S. Zaghrou, et al., Orbital motion around the collinear libration points of the restricted three-body problem, *JAMCS*, **29** (2018), 1–16.
19. R. D. Tiwary, V. K. Srivastava and B. S. Kushvah, Computation of three-dimensional periodic orbits in the Sun-Earth system, *Phys. Astron. Int. J.*, **2** (2018), 98–107.
20. R. D. Tiwary, B. S. Kushvah, Computation of halo orbits in the photogravitational Sun-Earth System with oblateness, *Astrophys. Space Sci.*, **357** (2015), 73.
21. M. Delva, Integration of the elliptic restricted three-body problem with Lie series. *Celest. Mech. Dyn. Astron.*, **34** (1984), 145–154.
22. A. Hanslmeier, Application of Lie-Series to regularized problem in celestial mechanics, *Celest. Mech. Dyn. Astron.*, **34** (1984), 135–143.
23. A. Mittal, Md. S. Suraj and R. Aggarwal, The analysis of periodic orbits generated by Lagrangian solutions of the restricted three-body problem with non-spherical primaries, *New Astron.*, **74** (2020), 101287.
24. H. Peng, J. Zhao, Z. Wu, et al., Optimal periodic controller for formation flying on libration point orbits, *Acta Astronaut.*, **69** (2011), 537–550.
25. H. Peng, X. Jiang and B. Chen, Optimal nonlinear feedback control of spacecraft rendezvous with finite low thrust between libration orbits, *Nonlinear Dyn.*, **76** (2014), 1611–1632.
26. Y. Jiang, Equilibrium points and orbits around asteroid with the full gravitational potential caused by the 3D irregular shape, *Astrodyn.*, **2** (2018), 361–373.
27. X. Wang, H. Peng, A symplectic moving horizon estimation algorithm with its application to the Earth-Moon L_2 libration point navigation, *Astrodyn.*, **3** (2019), 137–153.
28. F. A. Abd El-Salam, D. A. Katour and M. O. Shaker, On the equilibrium point L_1 in the photogravitational oblate relativistic R3BP with application on Sun-planet systems, *Eur. Phys. J. Plus*, **130** (2015), 54.
29. F. A. Abd El-Salam and S. E. Abd El-Bar, On the collinear point L_3 in the generalized relativistic R3BP, *Eur. Phys. J. Plus*, **130** (2015), 234.
30. F. A. Abd El-Salam and S. E. Abd El-Bar, The photogravitational and oblateness effects in the location of collinear point L_2 in the relativistic R3BP, *Sylwan*, **158** (2014), 126–149.

